# ME 451: Final Evaluation

# Design and Fabrication of Throat Surgery Holder and Retractor

BTP Group Number 23

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DEPARTMENT OF MECHANICAL ENGINEERING IIT KANPUR

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# The Problem and the Solution

## Problem:

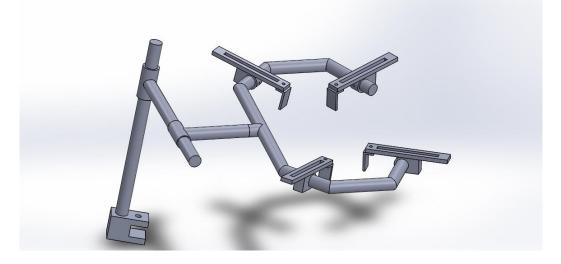
Rural community health centres in India are facing **83 per cent shortage of surgeons** [1]. Moreover, currently at least two surgeons are required to perform a surgery where the second surgeon's job is mostly to hold the tissue and flesh at the incision using hand-held retractors.

In this project, our objective is to reduce the skilled man-power required during operations by designing a mechanical system with the capabilities of holding and retracting skin and tissue at the incision during surgeries and thereby providing a more efficient and obstruction-free surgical site for surgeons.

### **Proposed solution:**

We propose a multitasking mechanical framework to assist the surgeon by holding and retracting skin and tissues at the incision without manual support. The whole system with multiple degrees of freedom will be mounted on a stand which will be fixed with the patient's bed providing stability. Retractor blades will be designed with the ability to adjust according to body profile appropriately during surgery and hold the skin at incision.

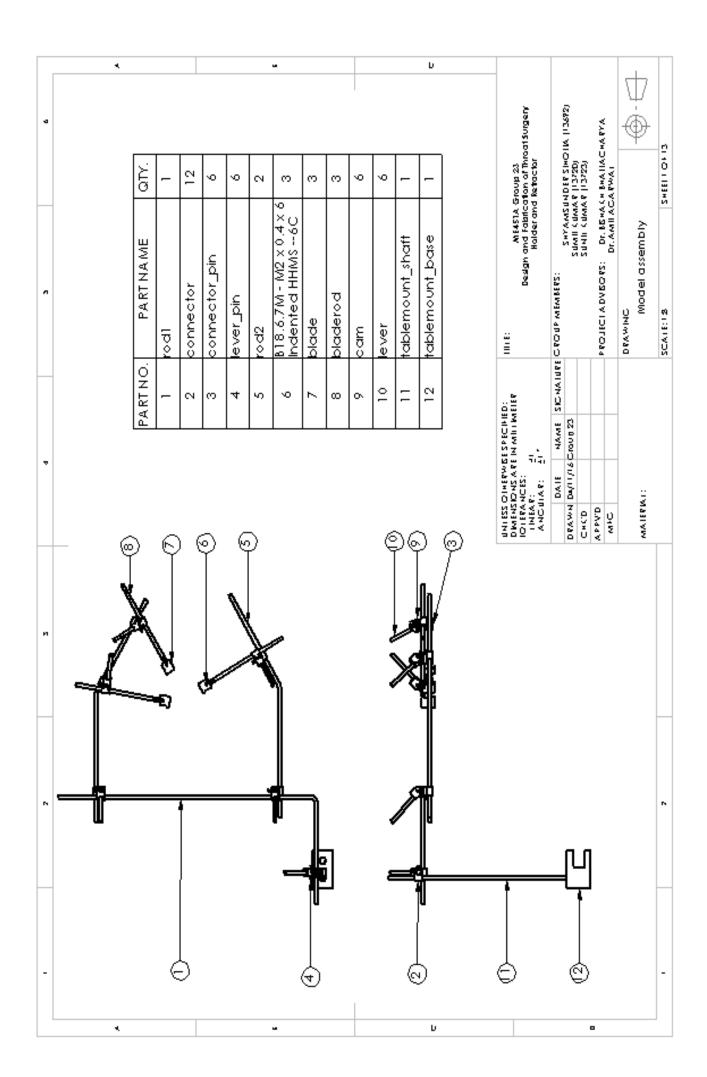
### **Conceptual Solution:**

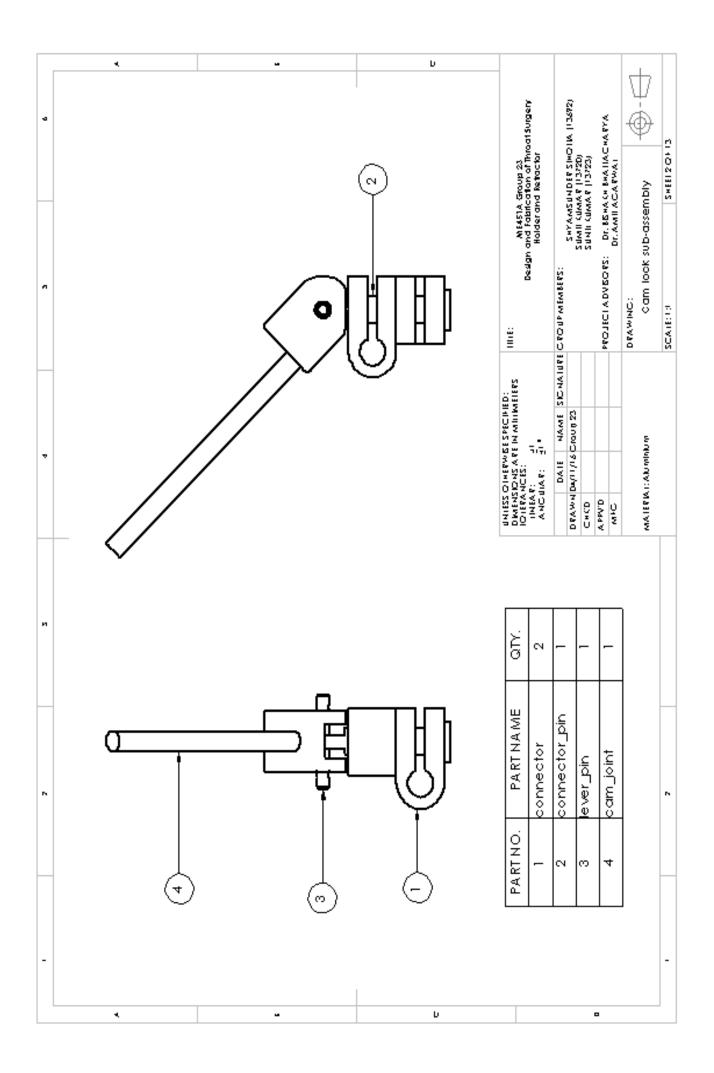


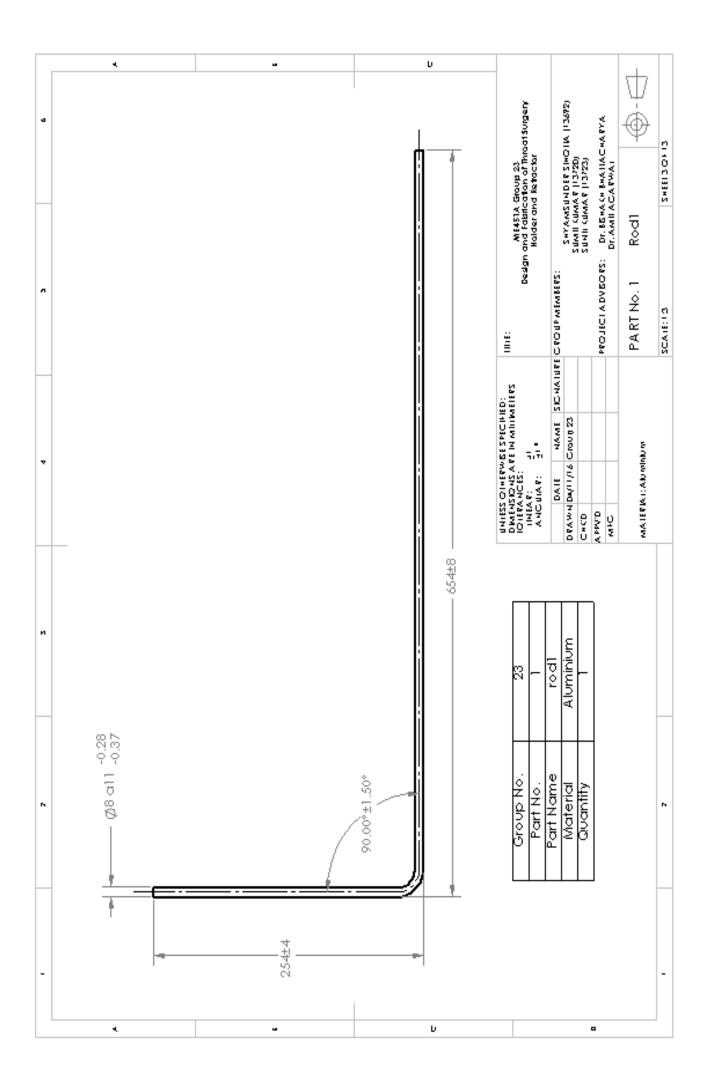
#### **Deliverable:**

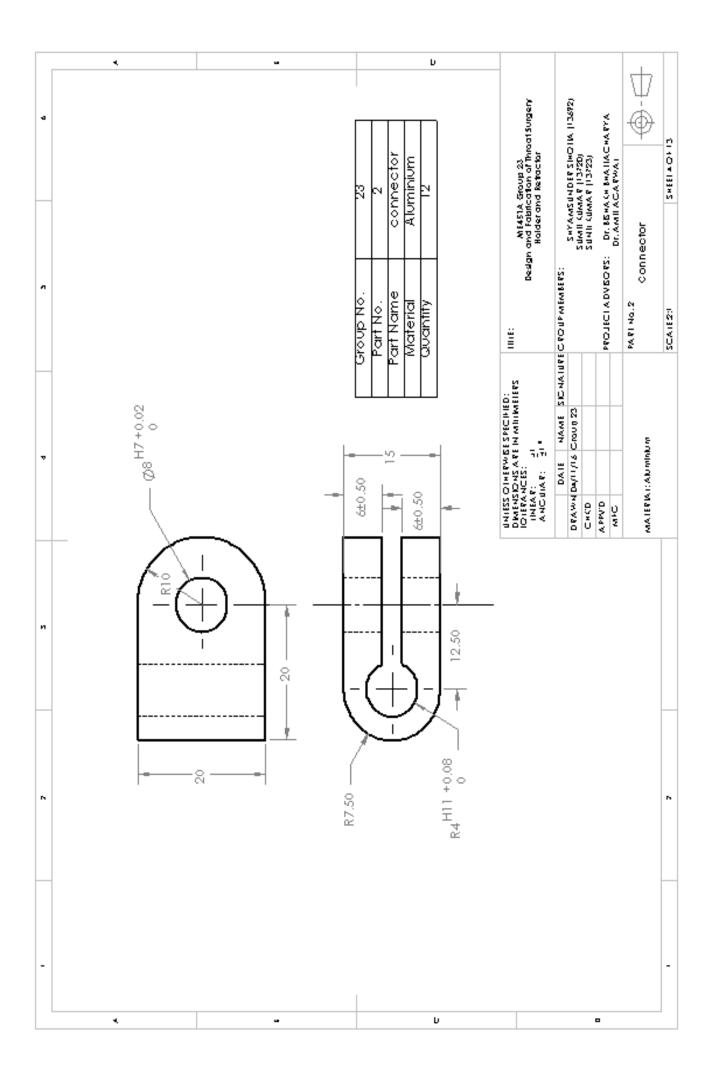
At the end of the manufacturing, we will be present a working laboratory prototype of the complete retractor system. We will be display a simulation of the prototype which will be used by a single individual and thereby successfully reduce the man power required in the surgery.

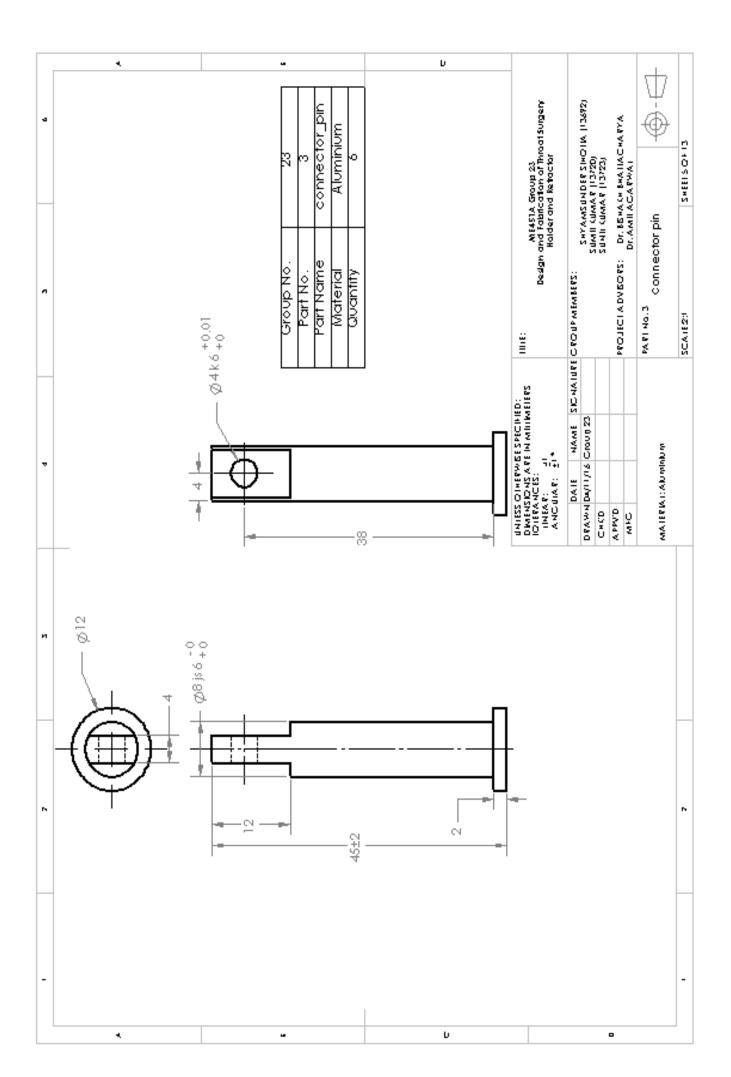
[1] http://scroll.in/article/756973/indias-community-health-centres-are-in-dire-need-of-more-specialists

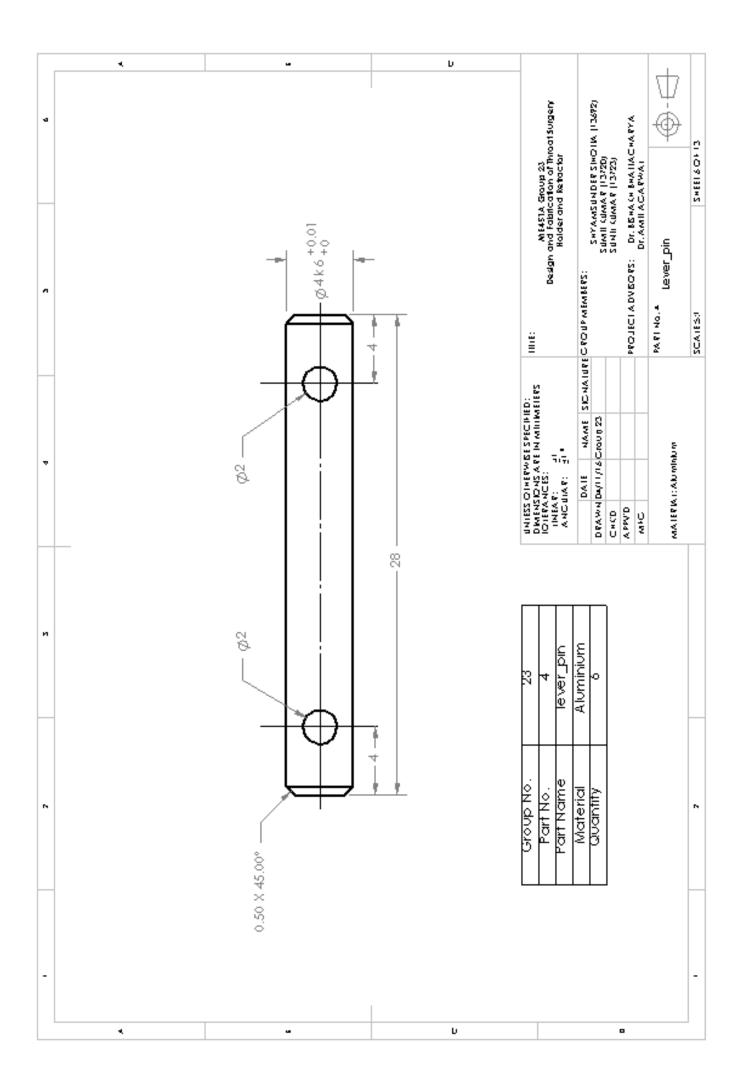


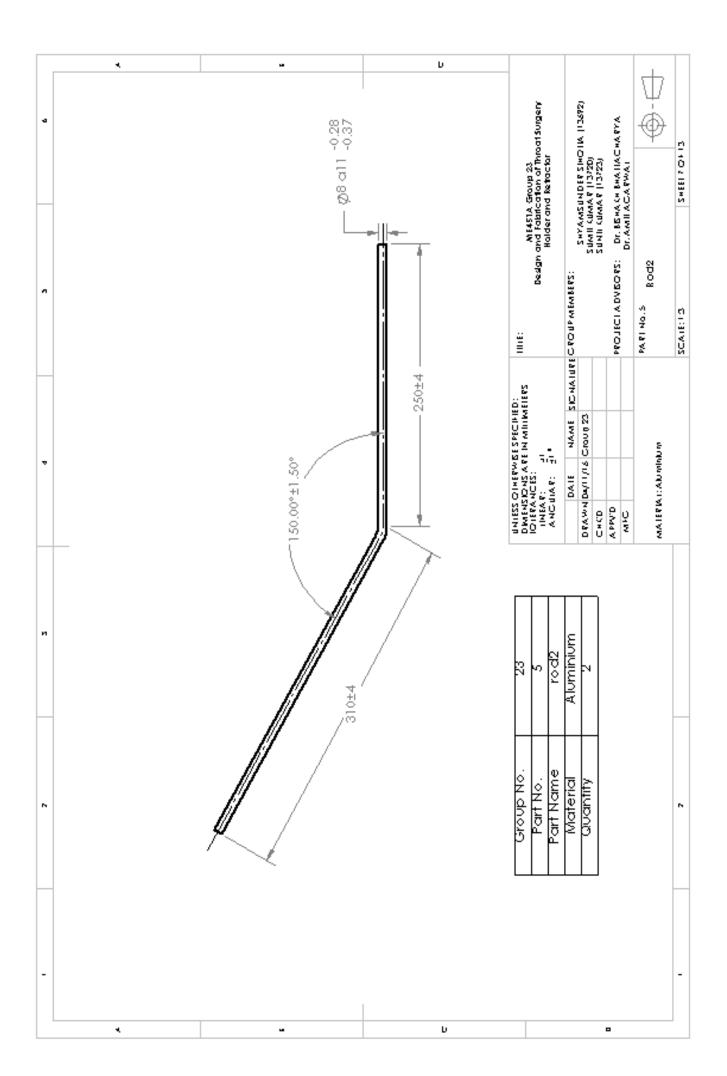


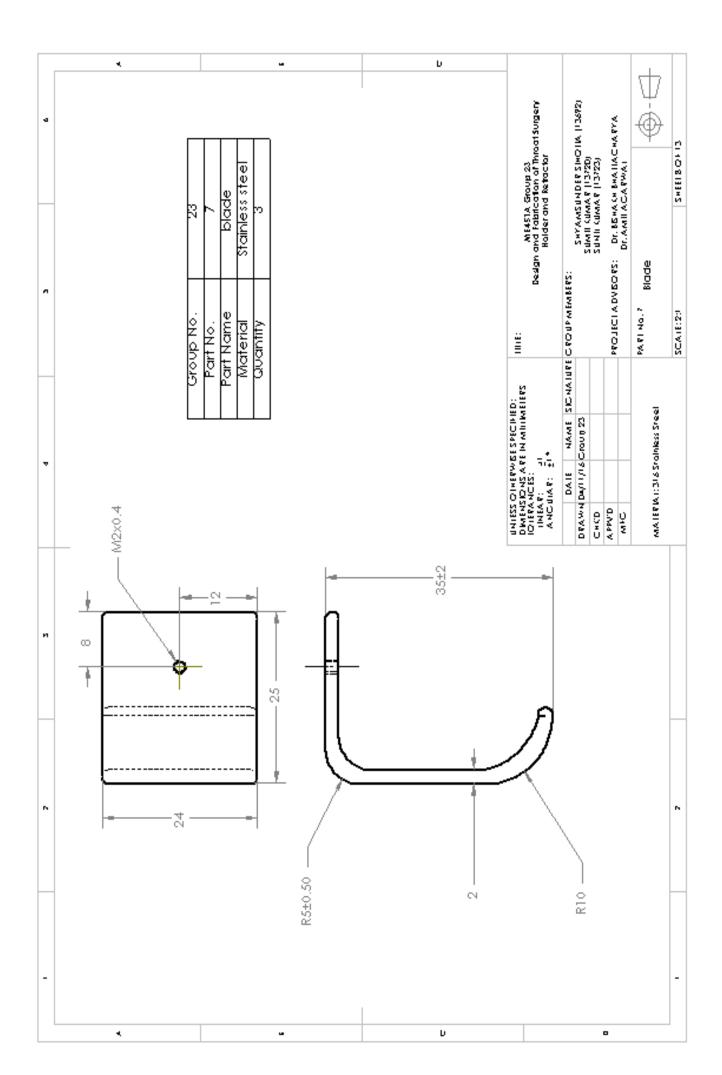


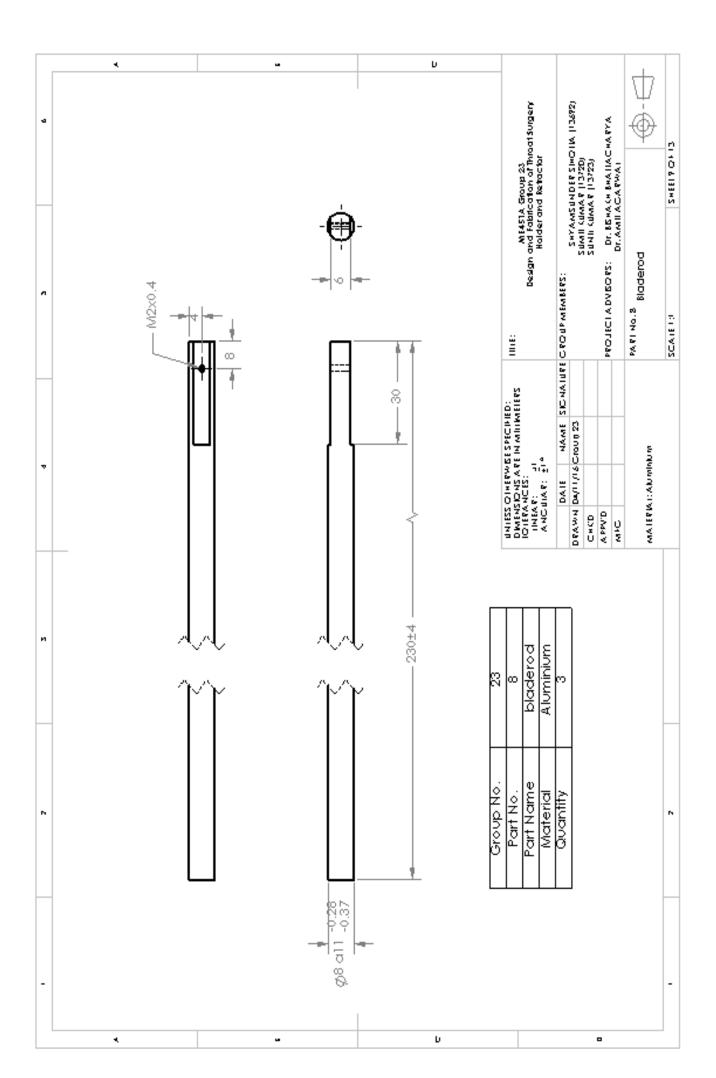


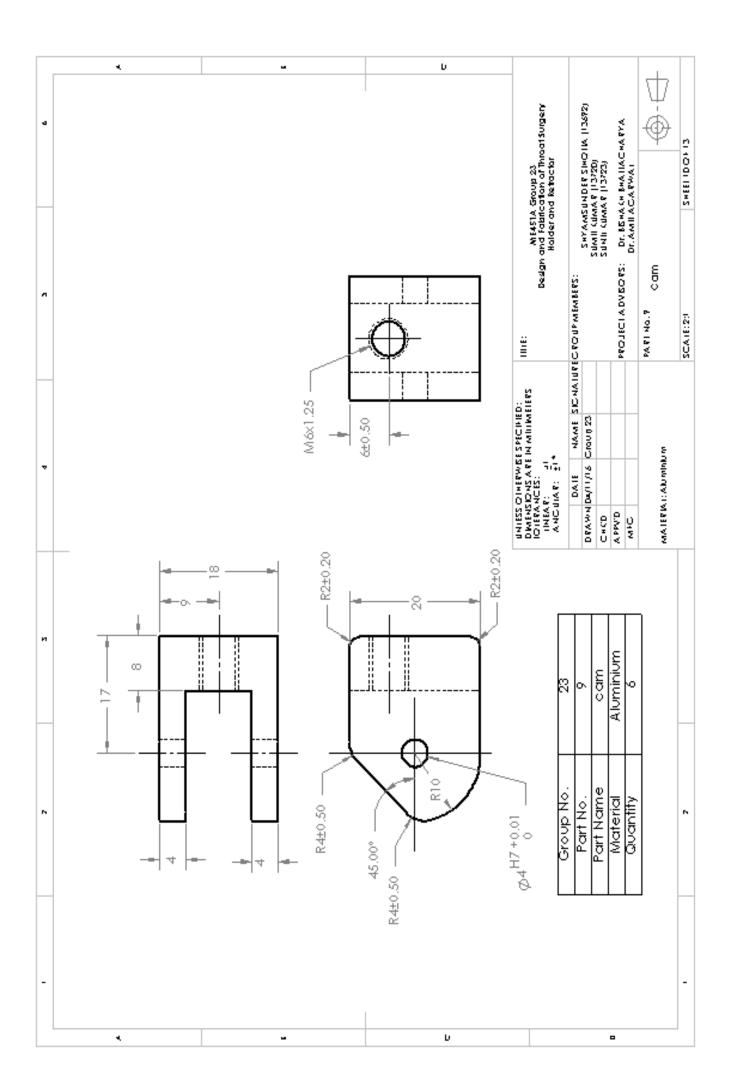


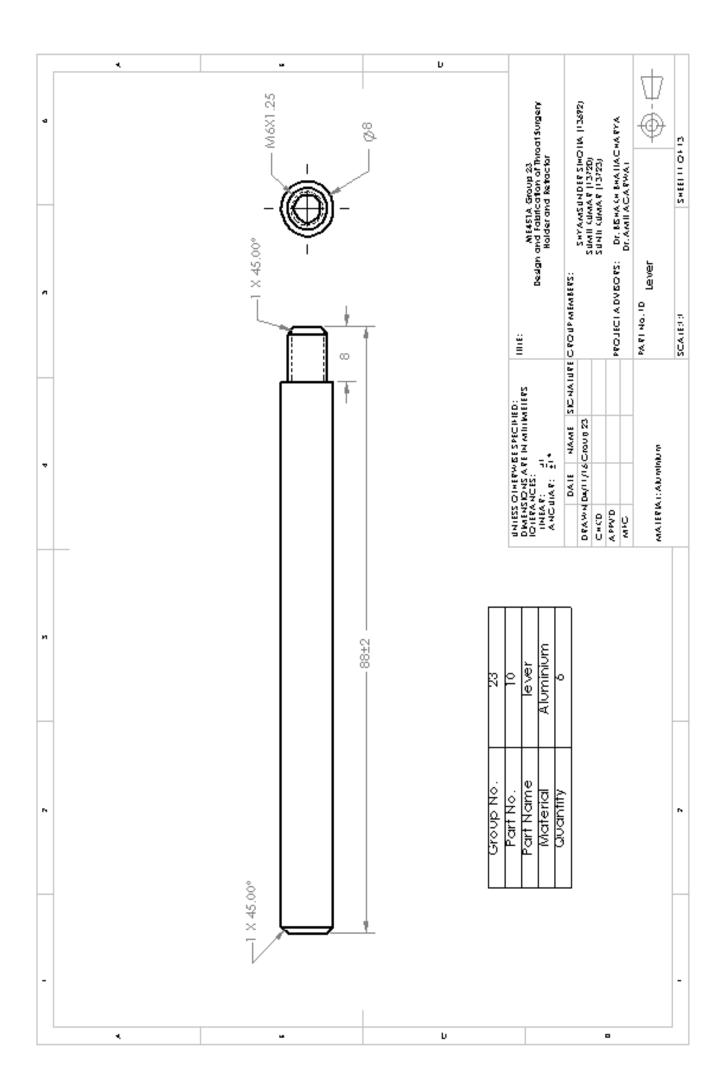


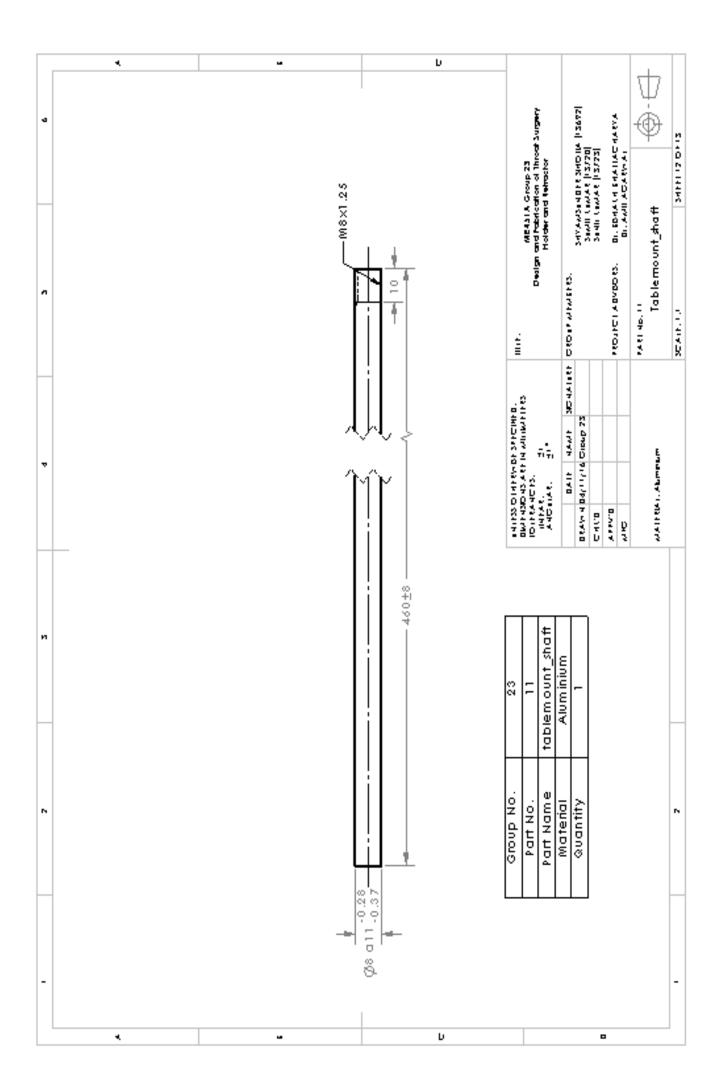


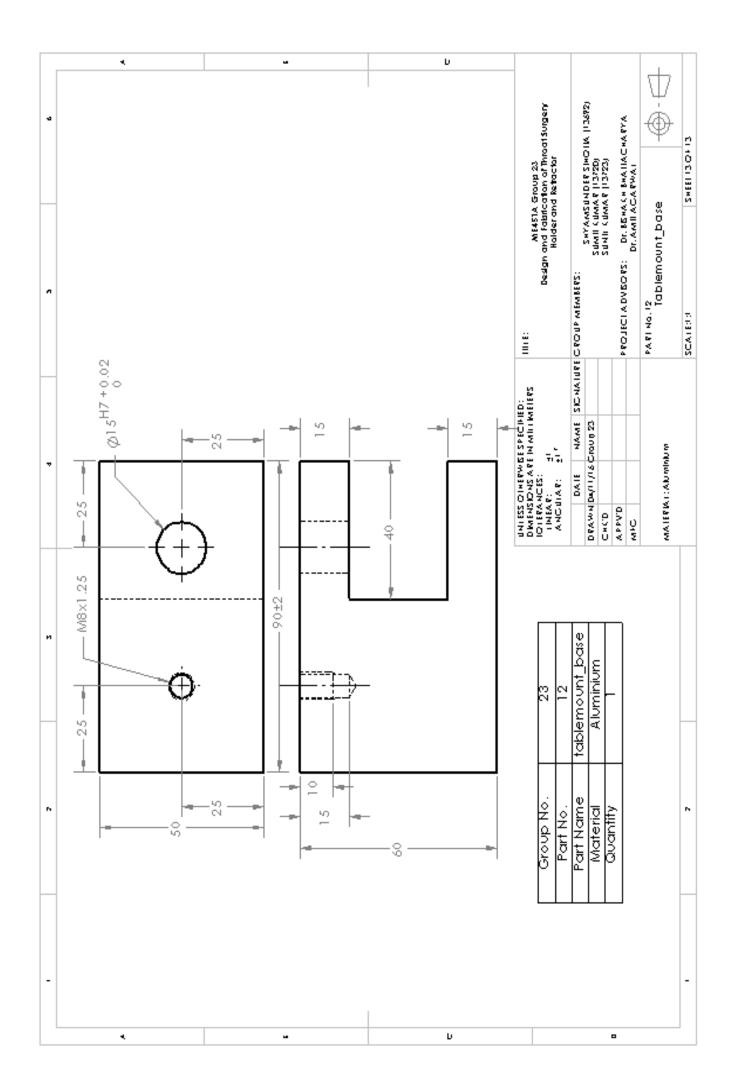












# **Final Design Specifications**

- Light weight: The complete retractor system weighs just 1.3 kg. This increases the usability of the product and also makes it portable. Also, with light weight it decreases the risks during surgery.
- Degrees of Freedom: The system has a total of 21 degrees of freedom providing it unhindered movement and useful in all variety of circumstances.
- Lifecycle: The fatigue testing results carried out with more than enough quantity of force shows 10<sup>6</sup> lifecycles before failure.
- Mechanical Stress Analysis: The mechanical stress analysis calculations clearly mark the system safe and not prone to failure with safety factors much higher than 1.
- Thermal Stress Analysis: Thermal analysis when conducted for the system made with surgical steel shows no sign of failure at any region.
- Surgical versatility: The system is versatile. With a change in the number and type of blades and proper alignment of rods, this system could even be used for performing surgeries at different locations and not just throat.

# Bill of materials

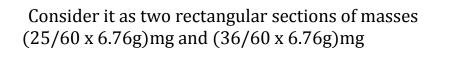
PART NO.	PART NAME	QTY.	Material	Major Dimensions (mm)
1	Rod 1	1	Aluminium	Ф8 х 910
2	Connector	12	Aluminium	30 x 20 x 15
3	Connector Pin	6	Aluminium	Ф12 х 45
4	Lever Pin	6	Aluminium	Ф4 x 32
5	Rod 2	2	Aluminium	Ф8 x 555
6	B18.6.7M - M2 x 0.4 x 6 Indented HHMS6C	3	Standard product (off-shelf)	M2 x 0.4 x 6
7	Blade	3	316 Stainless Steel	60 x 24 x 2
8	Blade Rod	3	Aluminium	Ф8 x 230
9	Cam	6	Aluminium	27 x 20 x 18
10	Lever	6	Aluminium	Ф8 х 88
11	Table mount Shaft	1	Aluminium	Ф8 x 460
12	Table mount Base	1	Aluminium	90 x 60 x 50

# **Design Calculations**

# Blade Rod Analysis (Stress Calculations)

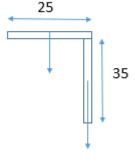
 $W_{blade} = M_{blade} \times g = (6.76 \times 10^{-3} \times 9.81) \text{ N}$   $W_{blade} = 0.066N$ 

 $W_{\mbox{\scriptsize blade}}$  will act at the center of mass of the blade



 $R_{cm} = \frac{\frac{25}{60}x \ 6.76x12.5 + \frac{36}{60}x \ 6.76x12.5}{6.76}$ 

 $R_{cm} = 19.79 mm$ 





Taking a conservative approach, consider the situation where whole blade hangs from the cam joint; i.e.

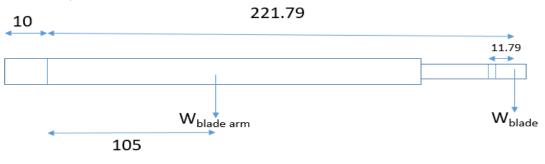
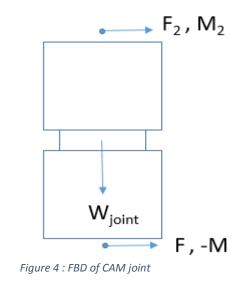


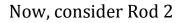
Fig. 3: FBD of blade arm

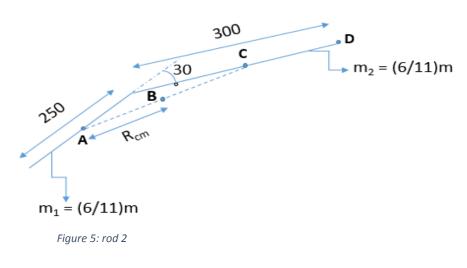
Let the forces and moment at support be  $\vec{F}$  and  $\vec{M}$ Then,  $\vec{F} = W_{\text{blade}} + W_{\text{blade arm}}$   $\vec{F} = \mathbf{0.37N}$   $\vec{M} = W_{\text{blade arm}} \ge 0.105 + W_{\text{blade}} \ge 0.22179$  $\vec{M} = \mathbf{0.046} \text{ Nm}$  [Direction is perpendicular to the blade arm]

Now, consider equilibrium of cam joint  $\vec{W}_{joint} = 0.56N$   $\vec{F}_2 = \vec{F} + \vec{W}_{joint}$   $\vec{F}_2 = 0.93N$ Now,  $\vec{M}_2 = \vec{M} = 0.046 \text{ Nm} [\vec{M}_{joint} + (-\vec{M}) = 0]$ 



# Arm Rod Analysis (Left Rod)





 $R_{cm} = \frac{m1 \times 0 + m2 \times r}{m1 + m2}$ 

 $r = \sqrt{125^2 + 150^2 - 2xa125x150xcos(150)}$ r = 265.71mm R<sub>cm</sub> = (6/11)r = 144.9mm

Hence, AB = 144.9mm BC = 120.81mm

Weight of Rod1 acts at point B.

again, taking a conservative approach, we take the case when blade arm is mounted at the extreme end of Rod1

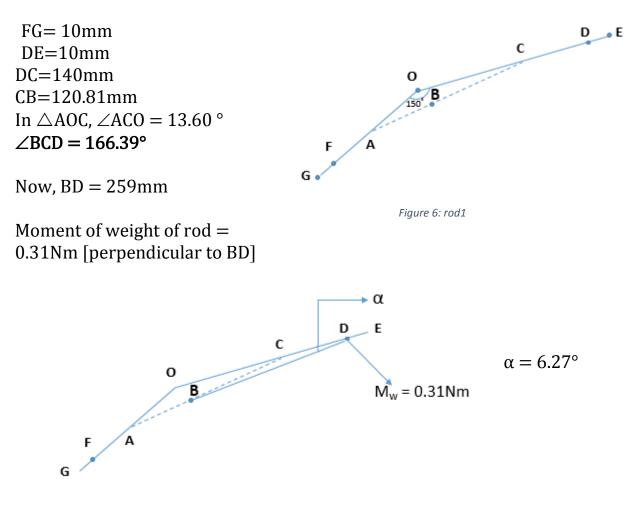
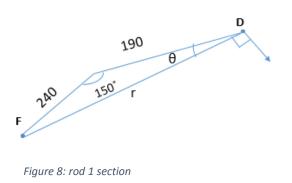


Figure 7: rod 1

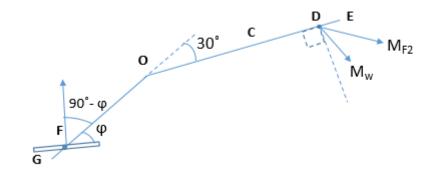


Let us find out moment due to force  $\vec{F}_2$  which acts at point F

r = 415.55mm $\theta = 16.78^{o}$ 

 $\vec{M}$ F2= 0.38 Nm

# Superposition of all moments at point D.



 $\phi$  is the angle between blade rod and Rod2 Take  $\phi = 30^{\circ}$ 

Figure 9: Moments acting on rod

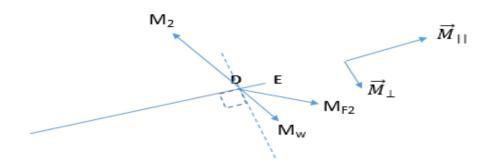


Figure 10: Net moment

Resultant,

$$\vec{M}_{||} = \vec{M}F2 \text{ x sin}(16.78^{\circ}) + \vec{M}_{rod} \text{ x cos}(6.27^{\circ})$$
$$\vec{M}_{||} = 0.143 \text{ Nm}$$
$$\vec{M}_{\perp} = \vec{M}F2 \text{ x cos}(16.78^{\circ}) + \vec{M}_{rod} \text{ x sin}(6.27^{\circ}) - \vec{M}_{2}$$
$$\vec{M}_{\perp} = 0.63 \text{ Nm}$$

$$\vec{F}_{net} = \vec{W}_{rod1} + \vec{F}_2 = 0.93 + 0.74$$
  
 $\vec{F}_{net} = 1.67N$ 

Bending Stress = 
$$\frac{Mc}{I}$$

 $= \frac{(0.63)(0.004)}{(\Pi/4)(0.004)^{4}}$ 

=12.53 MPa

Yield Strength = 240 MPa

Factor of safety (n) = 240/12.53 = 19.15

n>>1

Torsional Stress = 
$$\frac{(\vec{M}||,\text{net})r}{J}$$

$$=\frac{(0.143)(0.004)}{(\Pi/2)(0.004)^{4}}$$

= 1.422 MPa

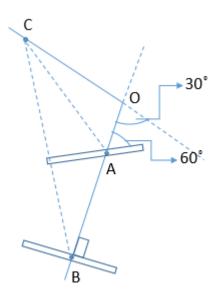
Tensile Strength = 290 MPa

Factor of safety (n) = 290/1.422 = 203

n>>1

## Arm Rod Analysis (Right Rod)

Now, considering the second rod having the two blade rods impended



Force and Moment at B and C are  $\vec{F}_{3}$ ,  $\vec{M}_{3}$  and  $\vec{F}_{4}$ ,  $\vec{M}_{4}$  respectively.

 $\phi 1$  and  $\phi 2$  have been taken to be 60° and 90° respectively. This is one of the various possible variations.

Figure 11: right rod

OB=240mm OC=290mm CB=512.10mm CA=325.26mm Also, take 0A=40mm

As calculated in previous section  $\vec{F}_3 = \vec{F}_4 = 0.93 \text{ N}$  $\vec{M}_3 = \vec{M}_4 = 0.046 \text{ Nm}$ 

Now,  $\angle BCO = 13.55^{\circ}$   $\angle ACO = 3.25^{\circ}$ Using these values to calculate moments through these forces we get,

 $\overrightarrow{M}$ F<sub>3</sub> = 0.48 Nm,  $\overrightarrow{M}$ F<sub>4</sub> = 0.30 Nm

Calculating net moment, setup and steps are same as in previous section, therefore

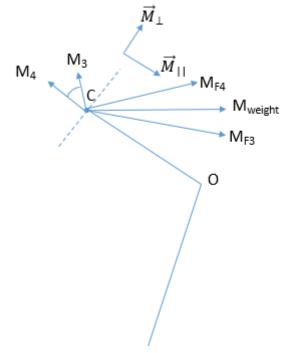


Figure 12: Moments acting on right rod

 $\overrightarrow{M}_{\text{weight}} = 0.31 \text{ Nm}$ 

 $\overrightarrow{M}_{\parallel} = 0.193 \text{ Nm}$ 

 $\overrightarrow{M}_{\perp} = 1.097 \text{ Nm}$ 

 $\vec{F}_{net} = 2x0.93 + 0.74$  $\vec{F}_{net} = 2.56 \text{ N}$ 

Bending Stress = 
$$\frac{Mc}{I}$$

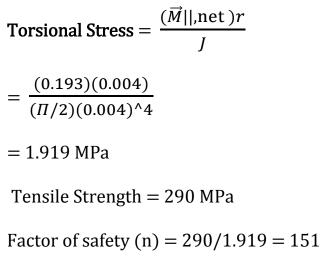
$$= \frac{(1.097)(0.004)}{(\Pi/4)(0.004)^{4}}$$

=21.82 MPa

Yield Strength = 240 MPa

Factor of safety (n) = 240/21.82 = 10.99

### n>>1



n>>1

# Analysis of Connecting Rod

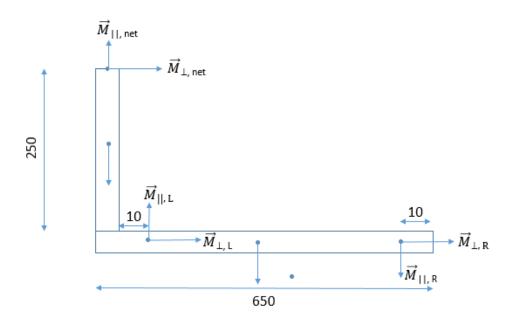


Figure 13: Rod 1

 $\vec{F}_1 = \vec{F}_{\text{net, L}} + \vec{W}_{\text{j}}$ 

$$\vec{F}_{1} = 1.67 + 0.56 = 2.23 \text{ N}$$

$$\vec{F}_{2} = \vec{F}_{\text{net},R} + \vec{W}_{1}$$

$$\vec{F}_{2} = 2.56 + 0.56 = 3.12 \text{ N}$$

$$\vec{W}_{1} = 120 \text{g} \times 0.001 \times 250/900$$

$$\vec{W}_{1} = 0.326 \text{ N}$$

$$\vec{W}_{2} = 120 \text{g} \times 0.001 \times 650/900$$

$$\vec{W}_{2} = 0.85 \text{ N}$$
At Point O,  

$$\vec{F}_{\text{net}} = \vec{W}_{1} + \vec{W}_{2} + \vec{F}_{1} + \vec{F}_{2}$$

$$= 120 \text{g} \times 0.001 + 2.23 + 3.12$$

$$\vec{F}_{\text{net}} = 6.55 \text{ N}$$

$$\vec{M}_{\text{IL,net}} = \vec{M}_{\text{IL,L}} - \vec{M}_{\text{IL,R}} + \vec{W}_{2} \times 0.325 + \vec{F}_{1} \times 0.001 + \vec{F}_{2} \times 0.640$$

$$\vec{M}_{\text{IL,net}} = \vec{M}_{\text{L,L}} + \vec{W}_{1} \times 0.125 + \vec{W}_{2} \times 0.250 + \vec{F}_{1} \times 0.250 + \vec{F}_{2} \times 0.250 + \vec{M}_{\perp,R}$$

$$\vec{M}_{\text{L,net}} = 3.95 \text{ Nm}$$
Bending Stress =  $\frac{(\vec{M} \perp \text{,net})c}{l}$ 

$$= \frac{(3.95)(0.004)}{(\pi/4)(0.004)^{\wedge}4}$$

$$= 78.58 \text{ MPa}$$
Yield Strength = 240 MPa  
Factor of safety (n) = 240/78.58 = 3.05 \text{ m} >1

Torsional Stress =  $\frac{(\vec{M}||, \text{net })r}{J}$ 

$$=\frac{(2.23)(0.004)}{(\Pi/2)(0.004)^{4}}$$

= 22.18 MPa

Tensile Strength = 290 MPa

Factor of safety (n) = 290/22.18 = 13.07

n>>1

# Analysis of Table Mount

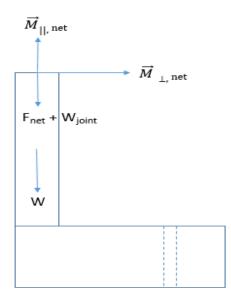


Figure 14: Table mount assembly

Bending Stress = 
$$\frac{(\vec{M} \perp, \text{net})c}{I}$$

 $=\frac{(3.95)(0.004)}{(\Pi/4)(0.004)^{4}}$ 

=78.58 MPa

Yield Strength = 240 MPa

Factor of safety (n) = 240/78.58 = 3.05

n>1  
Torsional Stress = 
$$\frac{(\vec{M}||, \text{net})r}{J}$$

$$=\frac{(2.23)(0.004)}{(\Pi/2)(0.004)^{4}}$$

= 22.18 MPa

Tensile Strength = 290 MPa

Factor of safety (n) = 290/22.18 = 13.07

n>>1

# Finite element stress analysis of cam

Force applied 100 N

1. Resultant stress

Model name: cam Study name: Static 1(-Default-) Plot type: Static nodal stress Stress1 Deformation scale: 1

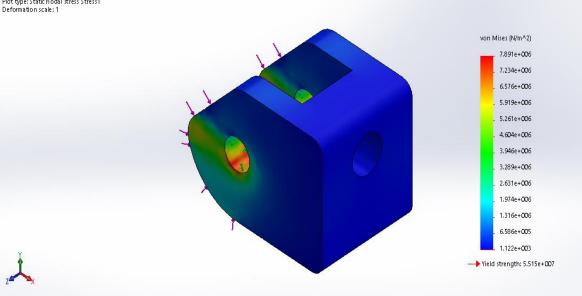


Figure 15: Resultant stress on cam

#### 2. Resultant deformation

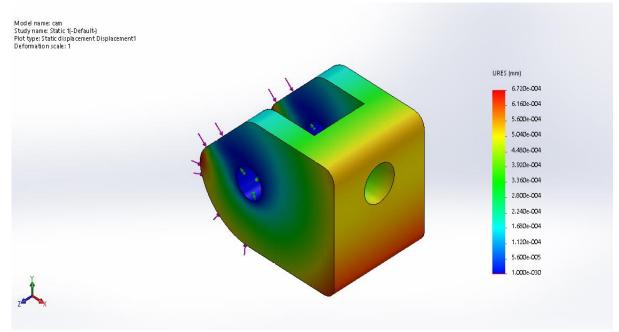


Figure 16: Resultant deformation in cam

#### 3. Resultant strain

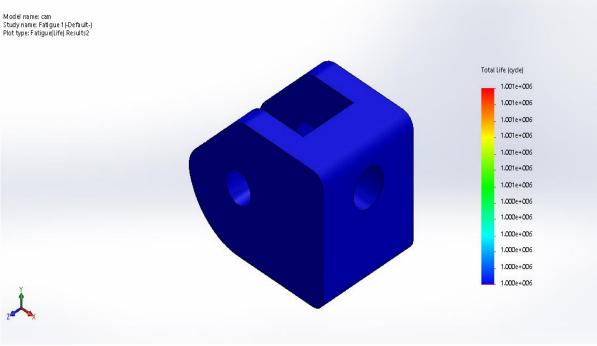


Figure 17: Resultant strain in cam

# Finite element stress analysis of cam lock assembly

#### 1. Stress results

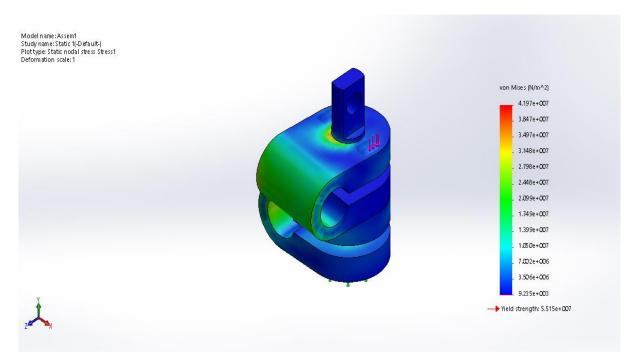


Figure 18: resultant stress on the sub-assembly

#### 2. Deformation results

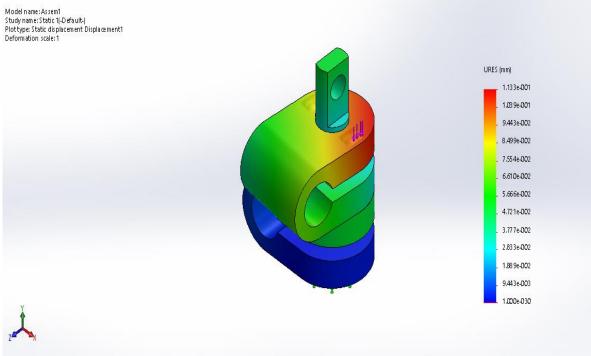


Figure 19: resultant deformation on the sub-assembly

#### 3. Strain results

Model name: Assem1 Study name: Fatigue 1(-Default-) Plottype: Fatigue(Life) Results2

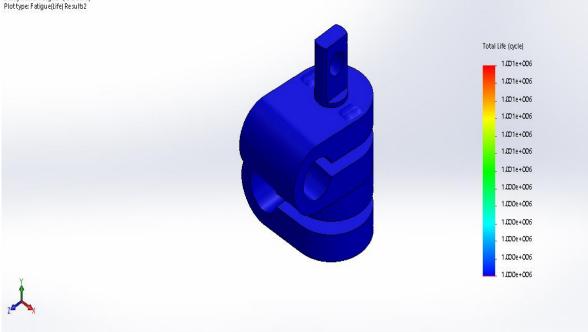


Figure 20: resultant strain on the sub-assembly

• Force exerted by flesh on the blade at the incision during surgery

 $F_v$  = force due to weight of the flesh above the blade

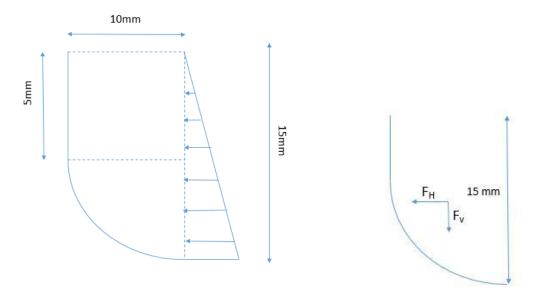


Figure 21: Forces acting on the blade surface due to human flesh

$$\begin{split} \textbf{F_v} &= \rho \textbf{gV} \\ \text{Where,} \\ \rho \text{ (average density of human flesh)} &= 1050 \text{ kg/m}^3. \\ & V &= [10x5 + \pi(10)^2/4] \text{ x } 24 \text{ x } 10^{-9} \text{ m}^3 = 3084.96 \text{ x } 10^{-9} \text{ m}^3 \\ & \text{g (acceleration due to gravity)} = 9.81 \text{ ms}^{-2} \\ \textbf{F_v} &= \rho \textbf{gV} = 1050 \text{ x } 9.81 \text{ x } 3084.96 \text{ x } 10^{-9} \\ \textbf{F_v} &= \textbf{0.032 N} \end{split}$$

$$\begin{split} F_{H} &= \text{horizontal force on the blade} \\ F_{H} &= \rho g h_{c} A \\ Where, h_{c} &= 7.5 \text{ mm} \\ A &= 15 \text{ x } 24 \text{ mm}^{2} = 360 \text{ mm}^{2} \\ F_{H} &= 1050 \text{ x } 9.81 \text{ x } 7.5 \text{ x } 360 \text{ x } 10^{-9} \\ F_{H} &= 0.028 \text{ N} \\ Thus, total force will be \end{split}$$

Figure 22: Resultant force

$$F = [F_{H^2} + F_{V^2}]^{1/2}$$
  
= [.028<sup>2</sup> + .032<sup>2</sup>]<sup>1/2</sup>

#### = $[0.001794]^{1/2}$ F = 0.042 N

### Finite element stress analysis of Blade

Force applied 0.042 N

1. Stress results

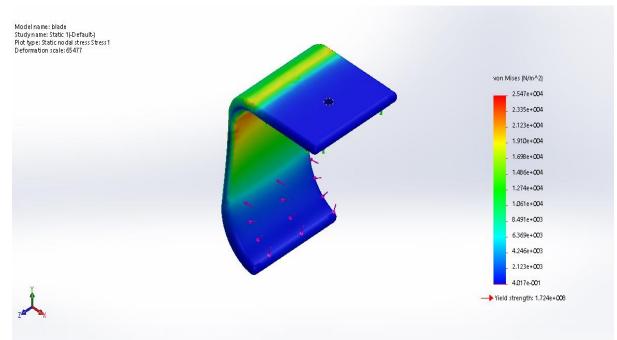


Figure 23: Resultant stresses on the blade

#### 2. Deformation results

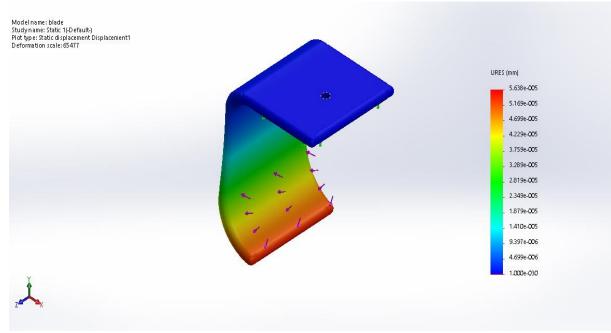


Figure 24: Resultant deformation

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#### Thermal Stress analysis

#### 1. Stress results

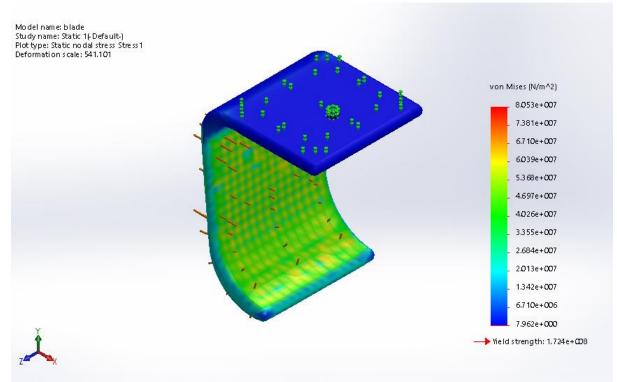


Figure 25: Thermal stress analysis of blade

#### 2. Deflection results

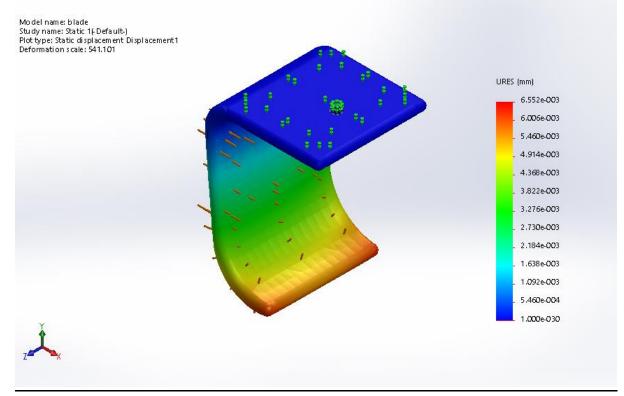


Figure 26: deflection on blade

# Parts designed using Design Calculations

PART NO.	PART NAME	QTY.
1	Rod 1	1
2	Connector	12
5	Rod 2	2
7	Blade	3
8	Blade Rod	3
9	Cam	6
11	Table mount Shaft	1

#### Summary

Out of the total number of 12 parts, design calculations have been done for 7 parts. Out of these 5 parts have written calculations and design calculations for 2 parts were through available softwares like Solidworks.

### Parts to be manufactured at IIT Kanpur:

OTY. PART PART NAME Material Major Dimensions NO. (mm) Φ8 x 910 Aluminium Rod 1 1 1 Aluminium 30 x 20 x 15 Connector 12 2 Connector Pin Φ12 x 45 Aluminium 6 3 Lever Pin Φ4 x 32 4 Aluminium 6 Rod 2 Aluminium Φ8 x 555 2 5 60 x 24 x 2 3 Blade 316 Stainless 7 Steel Aluminium Blade Rod Φ8 x 230 3 8 27 x 20 x 18 6 9 Cam Aluminium 10 Lever Aluminium Φ8 x 88 6 Table mount Shaft Aluminium Φ8 x 460 1 11 Table mount Base 12 Aluminium 90 x 60 x 50 1

All the parts of the assembly except part no. 6 will be manufactured at IIT Kanpur itself.

#### Off the shelf parts:

Part No. 6 is a standard part. It is available in the central workshop and will be used directly.

Part No.	Part Name	Material	Dimension	Qty.
6	B18.6.7M - M2 x 0.4 x 6 Indented HHMS6C	Mild steel	M2 x 0.4 x 6	3

### **Comments**

Manufacturing challenges:

The most intricate part of our design is the cam lock sub-assembly. It should be manufactured within the prescribed tolerances to enable the proper functioning of the entire model.

Cam profile has to be manufactured properly as per the proposed design. Moreover, the surface should be sufficiently smooth in order to avoid unnecessary noise while locking the joint.